

axial flow compressor, $R = \frac{V_a}{u} \tan \gamma_m$ where V_a is axial velocity, 'u' is blade speed and

 $\tan \gamma_{\rm m} = \frac{\tan \gamma_1 + \tan \gamma_2}{2}$

Important Note : 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.

(08 Marks)

The total power input at a stage in an axial flow compressor with symmetric inlet and outlet b. velocity triangles (R = 0.5) is 27.85 kJ/kg of air flow. If the blade speed in 180 m/s throughout the rotor, draw velocity triangles and compute inlet and outlet rotor blade angles. Assume axial velocity component to be 120 m/s. Would you recommend this compressor? (08 Marks)

15MR54

(06 Marks)

Module-3

- a. Briefly explain velocity compounding of impulse turbine. 5
 - Steam issuing from a nozzle to De-laval turbine with a velocity of 1000 m/s. The nozzle is b. 20° , the mass blade speed is 400 m/s. The blades are symmetrical, the mass flow rate = 1000 kg/hr, friction factor = 0.8, nozzle efficiency = 0.95. Calculate: (i) blade angles (ii) axial thrust (iii) work done per kg of steam (iv) power developed (v) blade efficiency. (10 Marks)

OR

- Define degree of reaction for a steam turbine and show that for a Parson's reaction turbine 6 a. (R = 0.5), degree of reaction is R = $\left(\frac{V_f}{2U}\right) \left[\operatorname{cut}\beta_2 - \operatorname{cot}\beta_1\right]$. (08 Marks)
 - The following data refers to a particular stage of a Parson's reaction turbine. Speed of b. turbine = 1500 rpm, mean diameter of rotor = 1 m, stage efficiency = 0.8, blade outlet angle = 20° , speed ratio = 0.7. Determine available isotropic enthalpy drop, in the stage.

(08 Marks)

Module-4

- Define the following terms with respect to hydraulic turbine: 7 a. (ii) Effective head
 - (iii) Hydraulic efficiency (i) Gross head (iv) Mechanical efficiency (v) Overall efficiency (08 Marks)
 - b. A double jet pelton wheel is required to generate 7500 KW when the available head at the base of nozzle is 400 m. The jet is deflected through 165° and the relative velocity of jet is reduced by 15% in parsing over the buckets. Determine: (i) Diameter of each jet (iii) Force exerted by jets in tangential direction. Assume $\eta_{gen} = 95\%$, (ii) Total flow $n_0 = 80\%$ and speed ratio (ϕ) = 0.47. (08 Marks)

OR

- What is a draft tube? What are important functions of draft tube and with neat sketch, show 8 a. different types of draft tube used in hydraulic power station. (08 Marks)
 - A Kaplan turbine working under a head of 15 m developed 7350 KW. The outer diameter of b. runner is 4 m and hub diameter is 2m. The guide blade angle at the extreme edge of runner is 30°. The hydraulic and overall efficiency of turbine are 90% and 85% respectively. If velocity of whirl at inlet is zero, then determine: (i) Runner vane angle at inlet and outlet (08 Marks) (ii) Speed of turbine.

Module-5

- Explain the following: a. (i) Pumps in series with neat sketch (ii) Pumps in parallel with neat sketch (iv) NPSH (08 Marks) (iii) Cavitation
- b. A CF pump delivers 1800 lpm against a total head of 20 m. Its speed is 1450 rpm. Inner and outer diameter of impeller are 120 mm and 240 mm respectively and diameter of suction and delivery pipe are both 100 mm. Determine blade angles β_1 , β_2 , if the water enters radially. Also find the power required to drive the pump. (08 Marks)

OR

a. Briefly explain the following with respect to CF compressor : 10

(i) Slip coefficient (μ)

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(ii) Power input factor (ψ)

(iv) Chocking

(08 Marks)

(iii) Surging b. A CF compressor running at 6000 rpm having an impeller tip diameter of 101 cm has following test data: mass flow rate = 25 kg/s, static pressure ratio = 2.12, pressure at inlet = 100 kPa, temperature = 28°C, mechanical efficiency = 0.97, find: (i) Slip coefficient (ii) Temperature of exit air (iii) Power input (iv) Power coefficient. (08 Marks)